

Attorney Docket No. 82649
Customer No. 23523

GPS ANTENNA FOR SUBMARINE TOWED BUOY

TO ALL WHOM IT MAY CONCERN:

BE IT KNOWN THAT DAVID F. RIVERA, employee of the United States Government, citizen of the United States of America, and resident of Westerly, County of Washington, State of Rhode Island has invented certain new and useful improvements entitled as set forth above of which the following is a specification:

MICHAEL P. STANLEY
Reg. No. 47108
Naval Undersea Warfare Center
Division, Newport
Newport, RI 02841-1248
TEL: 401-832-4736
FAX: 401-832-1231

2

3 **GPS ANTENNA FOR SUBMARINE TOWED BUOY**

4

5 **STATEMENT OF GOVERNMENT INTEREST**

6 The invention described herein may be manufactured and used
7 by or for the Government of the United States of America for
8 governmental purposes without the payment of any royalties
9 thereon or therefor.

10

11 **BACKGROUND OF THE INVENTION**

12 **(1) Field of the Invention**

13 The present invention relates to antennas and more
14 particularly to a global positioning system (GPS) antenna.

15 **(2) Description of the Prior Art**

16 In the field of GPS technology, GPS receivers are used to
17 determine the geographic location of the receiver by receiving
18 microwave radio signals from a group of earth-orbiting GPS
19 satellites. The geographic location of the receiver may be
20 computed by calculating its distance from each satellite as the
21 result of determining how long the signals take to travel from
22 the satellite to the receiver. Typically, a flat GPS antenna
23 element is utilized by GPS receivers to receive the signals
24 transmitted. In order for the GPS receiver to compute its

1 geographic location, the antenna element of the receiver must be
2 oriented to receive an acceptable level of the signals.
3 Optimally, the flattened surface of the GPS antenna element is
4 righted against the force of gravity such that a maximum surface
5 area of the antenna faces the satellites.

6 Present submarine communications with battlegroups or
7 satellites utilize surface antennas for a variety of
8 requirements including global positioning and communications.
9 The use of surface antennas typically interferes with the covert
10 operation of the submarine. For example, submarines obtaining
11 position fixes using GPS must raise a mast containing an antenna
12 which is oriented to receive the signals from the GPS
13 satellites. The problem is that raising a mast renders the
14 submarine vulnerable to either visual or radar detection,
15 especially if the mast is raised in coastal or littoral areas.

16 Additionally, antennas used on the ocean surface are
17 subjected to dynamic forces that act to cause the antenna to
18 pitch, yaw and sometimes roll with the vessel under varying sea
19 states. These antenna movements can easily re-orientate the
20 receiving element of the antenna resulting in reception
21 interruption. Varying sea states also cause a detuning effect
22 that result in degradation of the patch elements of conventional
23 GPS antennas. To minimize the effects of varying sea states,

1 the submarine must operate in a station keeping status or must
2 constantly adjust course headings.

3 One method of mitigating reception interruption of the
4 antenna is to orient the flattened surface of the antenna to
5 right itself or face "up" toward the sky irrespective of the
6 movement of its supporting structure. In Ham (U.S. Patent No.
7 6,292,147), an apparatus for maintaining a GPS antenna element
8 at a predetermined orientation is disclosed. The apparatus
9 includes a holder configured to support a GPS antenna element in
10 which the holder includes a rectangular frame as a receiving
11 portion of the dielectric substrate of antenna. The rectangular
12 holder pivots on an axis in relation to gravity to the
13 predetermined orientation even when the base structure to which
14 the holder is coupled changes its orientation. While the
15 disclosed reference allows a righting motion to the antenna
16 element, the movement of the righting motion is limited to
17 rotation around the axis of the pivot in which the rotation
18 provides only one degree of freedom.

19 It is well known in the use of gyroscopes and in the use of
20 compasses on ships, that a gimbal provides at least two degrees
21 of freedom for either attached device by allowing a pivoting
22 action on the axes of the gimbal in which the axes are rotatable
23 at angles to each other. For example, the pivoting and rotating
24 action of a gimbal used on a ship compensates for the roll and

1 the yaw of the ship as well as the pitch of the ship thereby
2 maintaining an accurate heading of a compass set in the gimbal.

3 As such, an improvement to the technology of GPS antennas
4 would be to incorporate the degrees of freedom of a gimbal with
5 a conformable GPS antenna in a manner that is suitable for use
6 on a vessel or towed array as well as for use in any other
7 situation that can require more than one degree of freedom in
8 which the degree of freedom is needed to maintain the righting
9 or facing up element of the antenna receiver. Such an
10 improvement along with any other suitable improvements to the
11 structure of the GPS antenna could act to minimize the reception
12 interruptions and the detuning effects caused by varying sea
13 states.

14

15 **SUMMARY OF THE INVENTION**

16 Accordingly, it is a general purpose and primary object of
17 the present invention to provide an apparatus with a Global
18 Positioning System (GPS) antenna that can obtain geographic
19 positioning data with minimal interruption when operating in
20 varying sea states.

21 It is a further object of the present invention to provide
22 an apparatus with an antenna that can transmit and receive
23 signal communications with minimal interruption when operating
24 in varying sea states.

1 It is a still further object of the present invention to
2 provide an apparatus with antenna that can be towed by a
3 submarine.

4 It is a still further object of the present invention to
5 provide an apparatus with antenna in which the construction is
6 simple and economical.

7 It is a still further object of the present invention to
8 provide an antenna capable of transmission at high frequencies
9 with minimal degradation.

10 It is a still further object of the present invention to
11 provide an antenna in which the construction is simple and
12 economical.

13 To attain the objects described, there is provided an
14 apparatus with a GPS antenna in which the antenna maintains a
15 receiving area that faces toward the sky or ocean surface. The
16 antenna is a hollowed frustum having a closed end at its
17 decreased diameter and an integral base ring surrounding an open
18 end at an increased diameter of the frustum. The antenna
19 includes a feed stem at the closed end extending as an internal
20 rod in the interior of the frustum. The opposite end of the
21 internal rod connects to a receiver plate in which the receiver
22 plate extends from the base ring toward and beyond a
23 longitudinal axis of the frustum.

1 For use in vessel operations or other applications that
2 require the receiver plate to face the sky or the ocean surface,
3 the antenna is supported by a gimbal. The gimbal is attachable
4 to the interior of a watertight container suitable for towing
5 horizontally on the ocean surface.

6 During operations, the pivoting of the antenna at the open
7 end in relation to the lower center-of-gravity of the frustum
8 shape of the antenna allows an enhanced swinging arc in relation
9 to the attached gimbal in that the body of the frustum moves by
10 gravity toward the axes of the gimbal. As such, the antenna
11 provides the righting or facing up of the open end of the
12 frustum and a facing up of the flattened surface of the attached
13 receiver plate thereby permitting enhanced reception by the
14 antenna. Furthermore, the antenna itself and not a holder of
15 the antenna provides the righting or facing up motion thereby
16 allowing a reduction in the amount of parts and a simplicity in
17 design.

18 During actuation of the antenna, the feed stem is
19 conductive to an energized feed source. Radio-frequency energy
20 from the feed stem continues to the frustum with the energy
21 disbursing as a current distribution along the interior surface
22 of the frustum. The radio-frequency energy from the feed stem
23 also continues onto the receiver plate with the result of a
24 current distribution across the receiver plate. The differences

1 in phase and amplitude from the radiating surface of the
2 frustum, and the receiver plate contributes to a hemispherical
3 radiation pattern in the far field.

4 The hemispherical radiation pattern is advantageous because
5 when the antenna is placed on the ocean surface, the radiation
6 pattern in the air space above the ocean surface does not
7 contain nulls. As such, the radiation pattern in the air space
8 permits full directionalized reception from GPS satellites or
9 other signal emitting sources.

10 Furthermore, the antenna of the present invention reduces
11 the degradation and associated problems with detuning occurring
12 during various sea states. Specifically, the impedance matching
13 of the frustum shape and the components of the antenna control
14 the impedance influence of the detuning. Also, the structure of
15 the curved frustum shape removes the edges of a typical patch
16 antenna in which the edges of the typical patch antenna are
17 subject to degradation from detuning.

18 The above and other features of the invention, including
19 various and novel details of construction and combinations of
20 parts will now be more particularly described with reference to
21 the accompanying drawings and pointed out in the claims. It
22 will be understood that the particular devices embodying the
23 invention are shown by way of illustration only and not as the
24 limitations of the invention. The principles and features of

1 this invention may be employed in various and numerous
2 embodiments without departing from the scope of the invention.

3 4 **BRIEF DESCRIPTION OF THE DRAWINGS**

5 A more complete understanding of the invention and many of
6 the attendant advantages thereto will be readily appreciated as
7 the same becomes better understood by reference to the following
8 detailed description when considered in conjunction with the
9 accompanying drawings wherein:

10 FIG. 1 is a side view of the antenna of the present
11 invention;

12 FIG. 2 is a plan view of the antenna of the present
13 invention with the view taken from reference line 2-2 of FIG. 1;

14 FIG. 3 is an alternate plan view of the antenna of the
15 present invention with the view taken from reference line 3-3 of
16 FIG. 1;

17 FIG. 4 is a side view of the antenna of the present
18 invention with the antenna mounted on a gimbal positioned in an
19 antenna housing;

20 FIG. 5 is a cross-sectional view of the antenna housing
21 attached to a tow body with the view taken from reference line
22 5-5 of FIG.4; and

23 FIG. 6 is a three dimensional view of a radiation pattern
24 formed by the antenna of the present invention.

DESCRIPTION OF THE PREFERRED EMBODIMENT

Referring now to the drawings wherein like numerals refer to like elements throughout the several views, one sees that FIG. 1 depicts the antenna 10 of the present invention. The antenna 10 is preferably cast with a rigid thickness from phosphor bronze or beryllium copper with electrically conductive components attached or also cast as part of the antenna. Other commonly acquired materials resistant to corrosion in a sea environment or materials known to those skilled in the art may be used in forming the antenna 10.

The simplified structure of the antenna 10 comprises a hollowed frustum 12 having an open end 14 and a closed end 16 with a distance between the closed end and the open end being approximately $\lambda/9$, wherein λ is the free-space wavelength measured in meters. For GPS use, the free-space wavelength equals the center frequency of operation, [the square root of the multiplication of the GPS frequencies (1227 MHz, 1575 MHz)] divided by the speed of light. The sizing of the diameter of the frustum 12 as well as the sizing of the other components of the antenna 10 is based on the free-space wavelength thereby allowing the antenna to be sized at a substantial bandwidth for alternate functions such as receiving and transmitting signals from IRIDIUM satellites (1625 MHz).

1 For the open end 14 of the frustum 12 shown in FIG. 2, the
2 open end has a diameter "A" of $2\lambda/5$. An integral base ring 18
3 projects from the open end 14 parallel to a longitudinal axis 20
4 of the antenna 10 in which the longitudinal axis is preferably
5 perpendicular to the open end 14 and the closed end 16. The
6 base ring 18 includes a notch 22 to position a receiver plate 24
7 flush with the projection of the open end 14. The receiver
8 plate 24 extends from the notch 22 to and beyond the
9 longitudinal axis 20. The receiver plate 24 is generally
10 rectangular in shape from the flush with the notch 22 with the
11 rectangular shape having a nominal length "B" of $\lambda/3$ and a width
12 "C" that is approximately ten percent less than the length "B".

13 For the closed end 16 of the frustum shown in FIG. 3, the
14 closed end 16 has a diameter of $\lambda/5$. The closed end 16 includes
15 a feed stem 30 shielded by an extension 31 of the frustum 12.
16 The feed stem 30 extends as an internal rod 32 in the cavity of
17 the antenna 10. See FIG. 1. For an optimum impedance match and
18 bandwidth to the antenna structure described above, the diameter
19 of the rod 32 is $\lambda/30$ with a length of $\lambda/10$ and a contact point
20 for the receiver plate 24 at $\lambda/11$ from the plate edge 34. The
21 depth of the cavity (noted above as the distance between the
22 open end 14 and the closed end 16), the size (the length "B" and
23 the width "C") of the receiver plate 24 and the size of the rod
24 32 determine the impedance at the feed stem 30, the radiation

1 pattern 36 of the antenna 10 and the bandwidth of the antenna
2 10.

3 Referring again to FIG. 2, the base ring 18 includes
4 attachment points 40, 42 in which the points allow the insertion
5 of a swivel axis or any other mechanical attachment to a gimbal
6 50, described below. As shown in FIG. 4 for the use of the
7 antenna 10 in submarine operations, the antenna 10 is supported
8 by the gimbal 50 attached to the interior of the watertight
9 container 52. The watertight container 52 is electrically
10 transparent polyethylene and is attachable to a tow body
11 54 (shown in FIG. 5) which can be towed by a submarine or other
12 vessel.

13 The pivoting at the attachment points 40, 42 of the antenna
14 10 in relation to the lower center-of-gravity of the frustum
15 shape of the antenna allows an enhanced swinging arc by gravity
16 (54) on the axis of the attachment points 40, 42 in relation to
17 the attached gimbal 50. The gimbal 50 in turn has a swinging
18 arc (56) on its own attachment points 58, 60; thereby providing
19 a righting movement for the antenna 10 on at least two axes. As
20 such, the antenna 10 provides the righting or facing up of the
21 open end 14 of the frustum and a facing up of a flattened
22 surface of the receiver plate 24 toward overhead satellites
23 thereby permitting enhanced reception by the antenna. The
24 antenna 10 is further unique in that the antenna itself and not

1 a holder of the antenna provides the righting or facing up
2 motion thereby allowing a reduction in moving parts and a
3 simplicity in design.

4 During actuation of the antenna 10, the feed stem 30 is
5 conductive to an energized feed source (not shown). Radio-
6 frequency energy from the feed stem 30 continues onto the
7 frustum 12 with the energy disbursing as a current distribution
8 along the interior surface of the frustum. The energy from the
9 feed stem 30 also continues to the receiver plate 24 by way of
10 the rod 32 with the result of a current distribution across the
11 receiver plate. The distribution of current amplitude and phase
12 from the surface of the frustum 12 and the receiver plate 24
13 contributes to a hemispherical radiation or beam pattern 36,
14 shown in FIG. 6. The hemispherical radiation pattern 36 is
15 advantageous because when the antenna 10 is placed on the ocean
16 surface, the radiation pattern in the air space above the ocean
17 surface (shown by the area 76 above the "x" and "y" axis) does
18 not contain nulls. As such, the radiation pattern in the air
19 space permits full directionalized reception from satellites.

20 Furthermore, the antenna 10 reduces the degradation and
21 associated problems with detuning occurring during with vary sea
22 states. Specifically, the impedance matching of the frustum 12,
23 the feed stem 30 and the rod 32 control the impedance influence
24 of the detuning. Also, the structure of the curved frustum 12

1 removes the edges of a typical patch antenna in which the edges
2 of the typical patch antenna are subject to detuning and quicker
3 degradation.

4 An additional feature of the present invention is that the
5 structural ratio (identified by the wavelength dimensioning
6 above) of the various components of the antenna 10 allows the
7 hemispherical radiation pattern 36 while maintaining the
8 compactness of the antenna 10. The compactness of the antenna
9 10 is advantageous for many reasons including detection
10 minimalization and reduced drag of the enclosing towing body.
11 In relation to conventional GPS antennas, the compactness of the
12 antenna 10 with its frustum 12 and receiver plate 24 does not
13 require a large ground plane in order to generate the
14 hemispherical radiation pattern 36.

15 In defining the compactness feature, the outer physical
16 boundary of the antenna 10 is based on the size and placement of
17 the open end 14 and the closed end 16 of the frustum 12. For
18 example, the diameters of the open end 14 and the closed end 16
19 are $2\lambda/5$ and $\lambda/5$ respectively with a distance of $\lambda/9$ between the
20 open end and the closed end. Any remaining structure of the
21 antenna 10 would be within a circumferential boundary created by
22 the above dimensions.

23 Furthermore, the all-metallic structure of the antenna 10
24 does not require a ceramic dielectric substrate yet allows

1 transmission and reception at a large instantaneous operating
2 bandwidth as exemplified by the antenna use with IRIDIUM and
3 global positioning signals described above.

4 Thus by the present invention its objects and advantages
5 are realized and although preferred embodiments have been
6 disclosed and described in detail herein, its scope should be
7 determined by that of the appended claims.